

# Measurement of the Elastic Constants of Carbon Fiber Board Based on Modal Analysis

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As a strategic new material with excellent performance, carbon fiber has attracted much attention in the wood structure industry. In this paper, the test modal and computational modal analysis methods are used to study the mode shape and modal parameters of carbon fiber board, calculate the elastic constants, *i.e.*, the elastic modulus, shear modulus, and Poisson's ratio, and analyze the coupling of the results. The conclusions show that the numerical results of the mode shape and elastic constants of the carbon fiber board obtained *via* the computational modal and test modal analysis are highly coupled. The first-order bending and first-order torsional modes of the carbon fiber board obtained *via* the two analysis methods are consistent, and the frequency error rate is less than 5%. The error rates of the elastic modulus and shear modulus are 0.7% and 7.8%, respectively. In addition, the research is conducive to strengthening the work of finite element computational modal analysis, and better promoting the application of non-destructive testing and quality grading of carbon fiber board. The above-mentioned optimization design work for improving the dynamic characteristics of carbon fiber board, and further research on the detection and optimization of the performance when the carbon fiber material is combined with wood, also has engineering application value for promoting the faster development of the wood structure market.

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*Keywords:* Carbon fiber board; Modulus of elasticity; Shear modulus; Poisson's ratio; Computational modal analysis; Test modal analysis

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## INTRODUCTION

The ancient buildings in China primarily adopted a wood frame structure, which is both light and beautiful as well as providing strong seismic resistance. However, due to the limitations of wood in terms of degradation resistance, fire prevention, as well as its old age, most of the existing ancient buildings are seriously damaged and need to be repaired (Yang *et al.* 2021). With the development of science and technology, more new materials and technologies have been applied to the protection of ancient buildings, which brings convenience to the protection of ancient buildings (Corbi *et al.* 2021; Liang *et al.* 2021; Xuan *et al.* 2021; Zhang *et al.* 2021). Since metal components are prone to corrosion when they are used to reinforce wood structures, the total additional weight will increase, which is not conducive to the reinforcement of wood structures (Dauletbek *et al.* 2021). Moreover, the traditional reinforcement method will cause damage to the building itself

(Ashraf *et al.* 2021; Liang *et al.* 2021; Xuan *et al.* 2021; Zhang *et al.* 2021). In recent years, with the continuous improvement of carbon fiber technology, carbon fiber materials have been used for the reinforcement of wood structures. Carbon fiber reinforcement technology has obvious advantages, primarily including its remarkable reinforcement effect, excellent degradation resistance and durability, no impact on the appearance of the structure, basically no change in the dead weight of the original structure, good fatigue resistance, *etc.* (Li *et al.* 2007; Gezer and Aydemir 2010). Therefore, wood structures are strengthened with carbon fiber, which includes carbon fiber cloth and carbon fiber board. Carbon fiber is a new material developed to meet the needs of aerospace, aviation, and other military departments. Its density is less than 1/4 of the density of steel, and its strength is 5 to 7 times greater than steel (Chen and Wang 2010). Compared with aluminum alloy structural parts, the weight reduction effect of carbon fiber can reach 20% to 40%, compared with steel parts. The weight reduction effect of carbon fiber can reach up to 60% to 80%. The specific strength of carbon fiber is 5 times higher than the specific strength of steel and 4 times higher than the specific strength of aluminum alloy (Yang *et al.* 2011). The specific modulus is 1.3 to 12.3 times greater than other structural materials. In addition, when the carbon fiber component is overloaded, it will cause a small amount of fiber fracture, and then the load will be transmitted and dispersed to other intact fibers so that the component will not lose its bearing capacity in a short time. These characteristics are enough to validate its excellent damage safety properties, and carbon fiber is easy to form in a large area (Wang *et al.* 2012). The manufacturing method of integral curing can be used, which greatly reduces the number of parts and fasteners, simplifies the production process, and shortens the production cycle. At present, carbon fiber is widely used in bridge reinforcement, aerospace applications, the military industry, medical technology, and other fields.

Since the 1980s, American researchers have developed carbon fiber cloth with superior properties, *e.g.*, light weight, high strength, a high modulus, corrosion resistant, and high fatigue resistant, which has been widely used in reinforcement projects. In 1997, China began to engage in research on carbon fiber reinforcement and its usage for the repair of civil building structures and completed the first engineering application of carbon fiber reinforced concrete structures in 1998. In 2014, ASTM standard C1557 (2014) proposed a standard test method for fiber tensile strength and elastic modulus. In terms of research on the elastic constants of carbon fiber, Duan *et al.* (2020) and Guruprasad *et al.* (2020) conducted nano-indentation tests on carbon fiber through the study of the indentation modulus. Using the numerical relationship between the indentation modulus and the elastic constant, the elastic modulus, in-plane shear modulus, and Poisson's ratio were determined through reverse analysis. Mesquita *et al.* (2021) used the fiber mechanical properties obtained *via* automated single fiber tensile test equipment to convert the load displacement diagram into a stress-strain diagram after performing compliance calibration, from which the elastic modulus and strength were extracted. Kim *et al.* (2009) calculated the in-plane shear modulus of composites using a simple relationship composed of the tensile modulus and Poisson's ratio. However, this method is only applicable to materials with a minimal fiber volume fraction that can be assumed to be isotropic. Choi *et al.* (2019) used digital image correlation technology to analyze unidirectional carbon fiber reinforced composites through three shear test methods, *i.e.*, in-plane  $\pm 45^\circ$ , Iosipescu shear test, and V-notch track shear test, and accurately measured the in-plane shear modulus of unidirectional carbon fiber. In order to explore whether carbon fiber and wood can be better integrated and further improve material properties, Novosel *et al.* (2021) studied the effect of implants

made of carbon fiber reinforced polymer (CFRP) on improving the bending stiffness of laminated oak model beams in two directions. The four-point bending method was used to evaluate the stiffness, bending strength and overall ductility of the elastic area. It was shown that the reinforcement effect was best when one or two layers of carbon fiber with a width of 40 mm and 60 mm were implanted, and the effective stiffness could be increased by up to 94%.

Most carbon fiber materials have the same anisotropy as wood, but the strength is not the same. In order to balance the production cost and mechanical properties of carbon fibers, isotropic pitch-based carbon fibers (IPCFs) are considered to be one of the most ideal candidate materials. However, there still have been relatively few studies on the elastic constant testing of IPCFs. With respects to these studies, dynamic testing and computational modal analysis of the elastic modulus, shear modulus, and Poisson's ratio of isotropic pitch-based carbon fiber (IPCFs) were carried out *via* the modal analysis method. Through the analysis of the coupling between the test model and computational model, the correctness and reliability of the elastic constant test of carbon fiber were ensured. In addition, the non-destructive testing level of the carbon fiber board material and the optimization design level of the dynamic characteristics were improved. This approach has practical significance to promote the development of the wood structure construction industry and further study the value of wood-carbon fiber composites.

## EXPERIMENTAL

### Modal Test Analysis of Carbon Fiber Board

#### *Materials*

A piece of carbon fiber board was selected for this test. The model was T700, with the following specifications: dimensions of 430.5 mm (length) × 301.5 mm (width) × 1.88 mm (thick), with a density of 1604.4 kg/m<sup>3</sup>, and manufactured by Jiangsu Boshi Fiber Technology Co., Ltd. (Jiangsu, China). The manufacturing process primarily includes two stages, *i.e.*, precursor production and precursor carbonization. Cut into fixed-length carbon fibers by SMC (Sheet Molding Compound) equipment, and make SMC preforms with the pre-prepared resin paste, and then put them in a constant temperature room for thickening and curing for 24 hours. The obtained SMC is vacuum molded with a vacuum hot press at 130 °C and a pressure of 2 MPa. After the SMC is cured, CFRP can be obtained. The main components of the resin matrix include unsaturated epoxy resin, vinyl resin, methyl ethyl ketone peroxide, cobalt isooctanoate accelerator, internal mold release agent and other auxiliary agents.

#### *Instruments and accessories*

A CRAS vibration signal acquisition and analysis system, which included an AZ-308 data acquisition box, AZ-802 signal conditioning box and MaCars software, was manufactured by Nanjing Anzheng Software Engineering Co., Ltd (Nanjing, China). The JF-LC-1003 force hammer, with a sensitivity of 2.07 mV/kN, a measurement range of 0 kN to 30kN, a hammer head mass of 350 g, a hammer handle length of 280 mm, and a rubber buffer head mass of 69.1 g, was manufactured by Yangzhou Jufeng Technology Co., Ltd. (Yangzhou, China). In addition, a TES-1350A sound level meter was used.

### Test method and principle

Based on the dynamic theory, the grid was divided on the free plate specimen of the carbon fiber board, with 6 equal parts in the length direction and 4 equal parts in the width direction. The single point vibration pickup and single point step excitation (SISO) methods of fixed measurement and moving excitation were adopted, and the force hammer excitation method, *i.e.*, the sound level measurement point, was fixed (point 7), so that the specimen produces transverse free vibration. The sound level meter was used to receive the sound signal, and the sound signal was transformed into an electrical signal through the vibration signal acquisition and analysis system. After signal amplification, low-pass filtering, spectrum identification, and an orthogonal test, the modal parameters and mode shapes of the carbon fiber specimens were finally obtained, and the first-order bending frequency of the free plate specimens were also obtained (Wang *et al.* 2012, 2014, 2016a).

When the test plate vibrates laterally, the relationship between the elastic modulus (E) and the first-order bending frequency ( $f_b$ ) is shown in Eq. 1,

$$E = 0.9462 \times \frac{\rho f_b^2 l^4}{h^2} \quad (1)$$

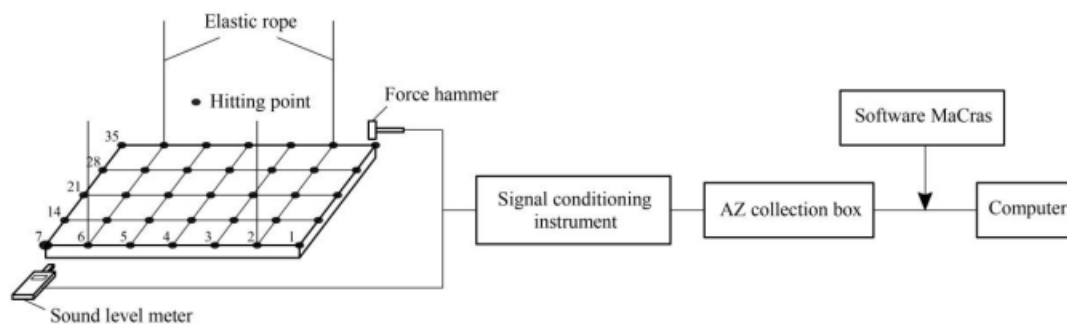
where  $\rho$  is the density of the specimen ( $\text{kg/m}^3$ ),  $f_b$  is the first-order bending frequency (Hz),  $l$  is the length of the specimen (m), and  $h$  is the thickness of the specimen (m) (Gao *et al.* 2013).

When the test plate vibrates laterally, the relationship between the shear modulus (G) and the first-order torsional frequency ( $f_t$ ) is,

$$G = \frac{\pi^2 \rho (l/2)^2 b^2 f_t^2}{7.5 \beta h^2} \quad (2)$$

where  $\rho$  is the density of the specimen ( $\text{kg/m}^3$ ),  $f_t$  is the first-order torsional frequency (Hz),  $l$  is the length of the specimen (m),  $h$  is the thickness of the specimen (m), and  $b$  is the width of the specimen (m) (Gao *et al.* 2016).

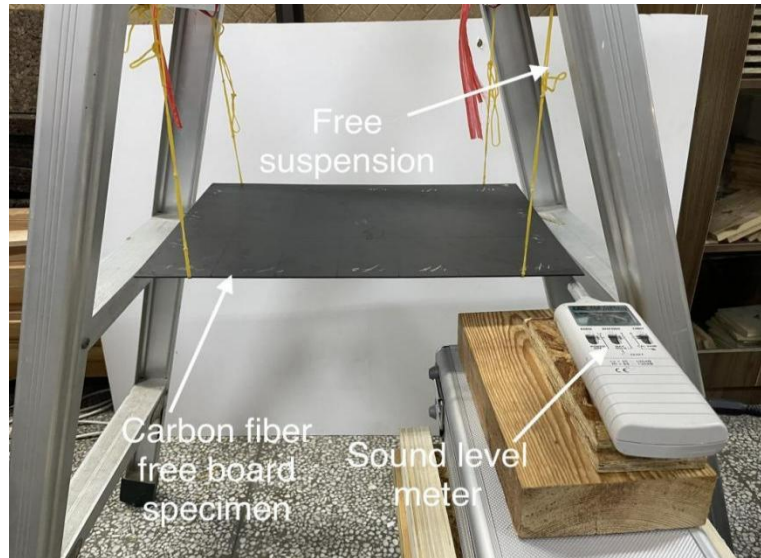
The system block diagram of the carbon fiber board free plate specimen is shown in Fig. 1.



**Fig. 1.** Test block diagram of the free plate modal test

### Test steps

A multi degree of freedom linear dynamic test system for the carbon fiber free plate specimens was realized, as shown in Fig. 2.



**Fig. 2.** Free plate modal test device

The data model was established using MaCars software. The parameter setting adopted the negative trigger mode, with a trigger level of 5%, a voltage range of  $\pm 625$  mV, an analysis frequency of 200 Hz, and a FFT block length of 1024.

According to the actual size of the carbon fiber board specimen, the model was established in MaCars, and its length and width directions were divided into 6 equal parts and 4 equal parts, respectively.

The sound level meter was placed, and the excitation points were divided. The measurement channel was set to 2, and the number in the Z direction was 1. During the test, the sound level meter was placed at point 7, *via* means of fixed measurement, single point vibration pickup of moving excitation, and single point step-by-step excitation (SISO). Through the force hammer excitation, the sound level measurement point (point 7) was fixed, the measurement method of the excitation point was changed, and the frequency response function of the fixed measurement point to each excitation point was measured. The SISO frequency response function identification method was used to complete the modal parameter identification by fitting the frequency response function curve.

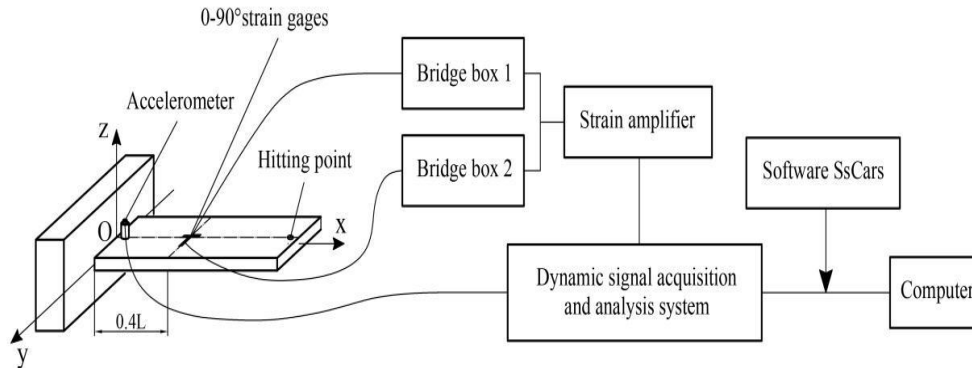
## Dynamic Test of Poisson's Ratio

### *Materials and instruments*

The dimensions of the carbon fiber board specimen were 490 mm (length)  $\times$  25.21 mm (width)  $\times$  3.51 mm (thickness), which was cut from the dimensions of 490.5 mm (length)  $\times$  301.5 mm (width)  $\times$  1.88 mm (thickness) board specimen, with a density of 1604.4 ( $\text{kg}/\text{m}^3$ ). The test instrument and its supporting parts are a set of CRAS vibration signal acquisition and analysis systems, which included an AZ-308 data acquisition box, AZ-802 signal conditioning box, and MaCars software (Nanjing Anzheng Software Engineering Co., Ltd (Nanjing, China). In addition, a rubber hammer, cantilever clamping device, and a CA-YD-125 acceleration sensor (with a sensitivity of 15.9  $\text{PC}/\text{m}\cdot\text{s}^2$  and measurement range of 0 hz to 20000 hz) were used. The sensitivity coefficient of the BX120-5AA strain gauge was  $2.08\% \pm 1\%$ , and the length and width of the strain grid were 5 mm and 3 mm, respectively.

### Test method and principle

The carbon fiber board specimen was fixed with the cantilever plate holding device, which had a holding length of 275 mm. The state of cantilever was completed, and an accelerometer was placed between the plate edge and the center line near the fixed end. Then, a 0° to 90° strain gauge was pasted on the cantilever plate, which was 0.4 L away from the clamping end. Next, the center line of the cantilever end of the specimen was hit with a rubber hammer to make the cantilever plate produce transverse vibration (Wang *et al.* 2015; Wang *et al.* 2016b). The test system block diagram is shown in Fig. 3.



**Fig. 3.** Test block diagram of the cantilever plate

The relationship between Poisson's ratio and the strain in the transverse and longitudinal directions of the specimen is shown in Eq. 3,

$$\mu_{ij} = -\varepsilon_i / \varepsilon_j \quad (3)$$

where  $\varepsilon_i$  is the absolute value of transverse strain and  $\varepsilon_j$  is the absolute value of longitudinal strain. Poisson's ratio is always equal to the absolute value of the ratio of the transverse strain to the longitudinal strain (Wang *et al.* 2017).

### Test steps

A strain gauge was pasted on the carbon fiber specimen on both sides, 0° parallel in the length direction of the specimen and 90° perpendicular in the length direction of the specimen. Then, the pasted specimen was fixed with the cantilever holding device and an accelerometer was placed near the center line of the fixed support end (Fig. 4). Then, the strain gauge was connected with the bridge box and the strain amplifier successively *via* a wire. The strain amplifier was connected with channel 1 and 2 of the dynamic signal acquisition and analysis system, in which one was connected with the 0° piece and two were connected with the 90° piece to realize a half bridge connection. Finally, the accelerometer was connected to the three channels of the dynamic signal acquisition and analysis system.

SsCars software was used for testing, with the following primary parameter settings: an analysis frequency of 1000 Hz, a FFT block length of 1024, and using the negative trigger mode with a trigger level of 25%.

Then, the cantilever plate was hit with the hammer, and its spectrum diagram was recorded (Fig. 5). The amplitude of the first-order bending frequency was replaced, corresponding to the 0° to 90° strain gauge, *i.e.*, the strain value, into Eq. 3 to obtain its Poisson's ratio, which are shown in Table 1.



Fig. 4. Cantilever plate test device



Fig. 5. Test spectrum of the cantilever plate

Table 1. Parameters and Test Results of the Cantilever Plate Specimen

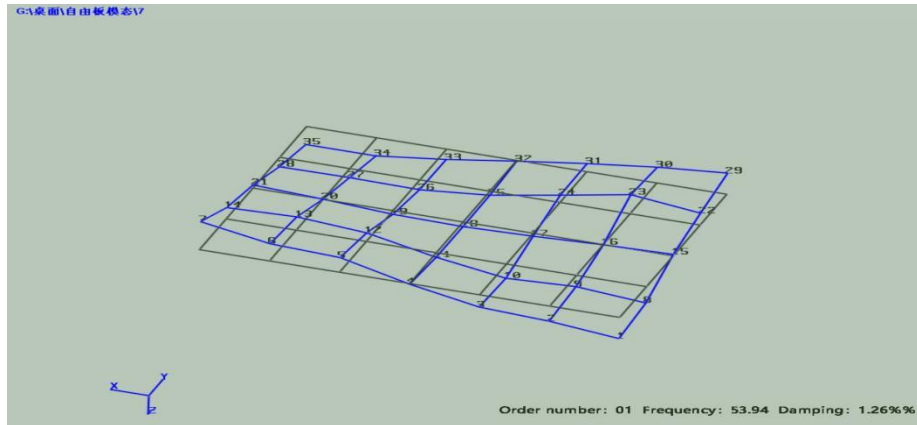
Order Number	Length (mm)	Width (mm)	Thickness (mm)	Density (kg/m <sup>3</sup> )	$\epsilon_{0^\circ}$	$\epsilon_{90^\circ}$	$\mu_{90-0^\circ}$
1	275	25.21	3.51	1604.4	20.48	9.51	0.46
2	275	25.21	3.51	1604.4	21.79	10.05	0.46
3	275	25.21	3.51	1604.4	26.59	12.01	0.46
4	275	25.21	3.51	1604.4	24.35	10.87	0.45
5	275	25.21	3.51	1604.4	22.08	10.40	0.47
Mean	275	25.21	3.51	1604.4	23.06	10.57	0.46



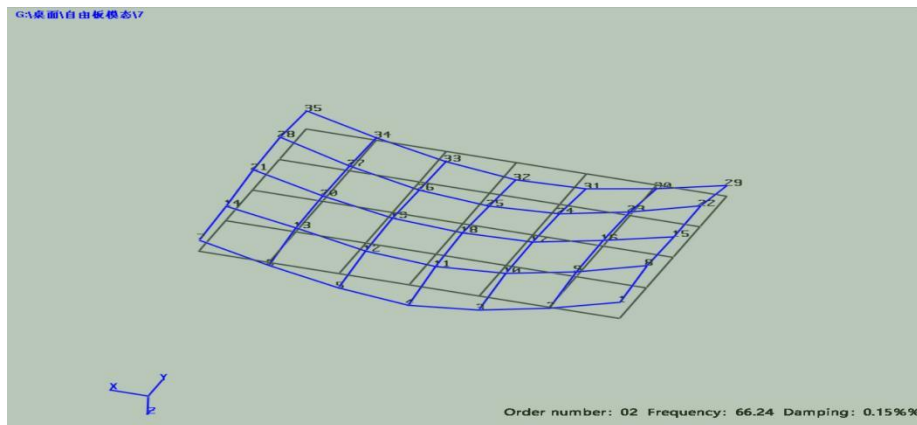
## RESULTS AND DISCUSSION

### Modal Test Results and Analysis

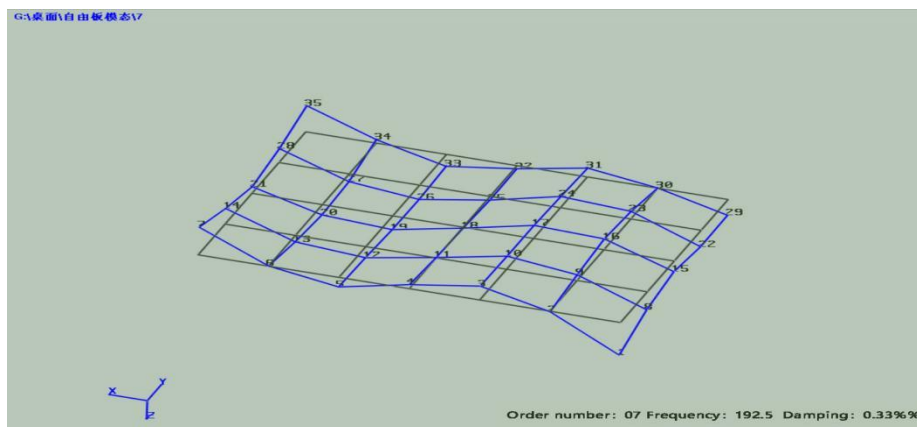
The mode shape of the carbon fiber specimen measured in this modal test are shown in Figs. 6 through 8.



**Fig. 6.** The first-order torsional mode of the free-plate



**Fig. 7.** The first-order bending mode of the free-plate



**Fig. 8.** The second-order bending mode of the free-plate



According to its mode shape, the first-order torsion mode and the first-order bending mode and their corresponding frequencies and damping ratios are shown in Table 2.

**Table 2.** Mode Natural Frequency and Damping Ratio

Content	First-order Torsional Mode	First-order Bending Mode
Mode frequency (Hz)	53.94	66.24
Damping ratio (%)	1.26	0.15

It can be seen from Table 2 that the first-order bending frequency and first-order torsional frequency obtained from the modal test of free plate specimen were 66.24 Hz and 53.94 Hz, respectively. As such, the elastic modulus ( $E$ ) and shear modulus ( $G$ ) were 64700 MPa and 22,200 MPa, respectively, as calculated from Eqs. 1 and 2, respectively.

### Computational Modal Analysis

In order to analyze the mode shapes and modal parameters of the carbon fiber board obtained by the above modal test method *via* the finite element theory and calculate the coupling of the elastic modulus and shear modulus, it is necessary to carry out a computational modal analysis of the carbon fiber board.

#### Procedure

Solidworks 2020 (Dassault Systèmes, Vélizy-Villacoublay, France) was used to model the specimens in this study, and the finite element modal analysis was performed in Workbench 19.1 (Ansys, Canonsburg, PA) for carbon fiber board material properties. The material properties are shown in Table 3.

**Table 3.** Material Properties of the Carbon Fiber in the Computational Model

Parameter	Density (kg/m <sup>3</sup> )	Length (mm)	Width (mm)	Thickness (mm)	The Elastic Modulus ( $E$ ) (MPa)	Poisson's Ratio ( $\mu_{ij}$ )
Specimen	1604.4	430.5	301.5	1.88	64731	0.46

The carbon fiber specimen was meshed, the node of the specimen was 1576, and the sensitivity of the mesh size was analyzed. The maximum mode calculation was set to order 20, free and unconstrained boundary conditions were set before calculation.

#### Computational modal results and analysis

The first-order torsional mode, first-order bending mode, and second-order bending mode of the specimen calculated *via* the finite element method are shown in Figs. 9 through 11, respectively. The calculated values are shown in Table 4

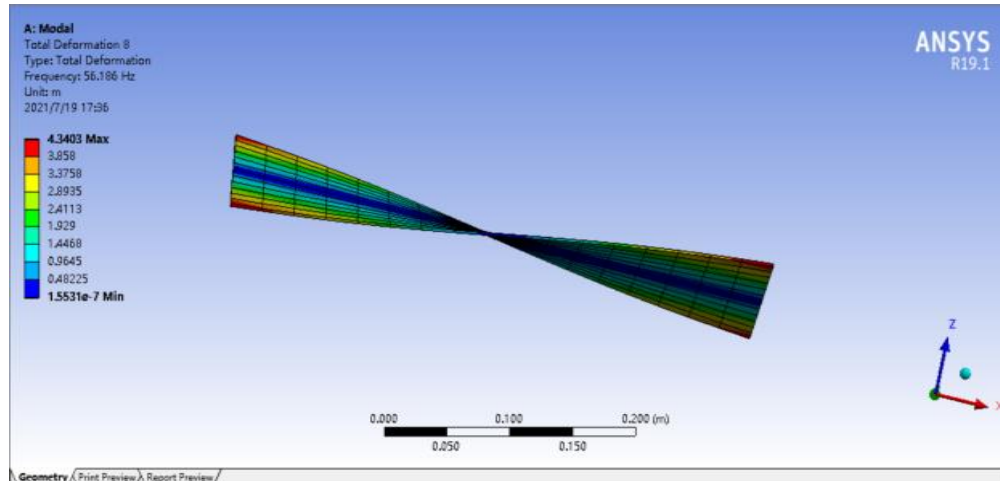


Fig. 9. The first-order torsional mode of the computational modal analysis

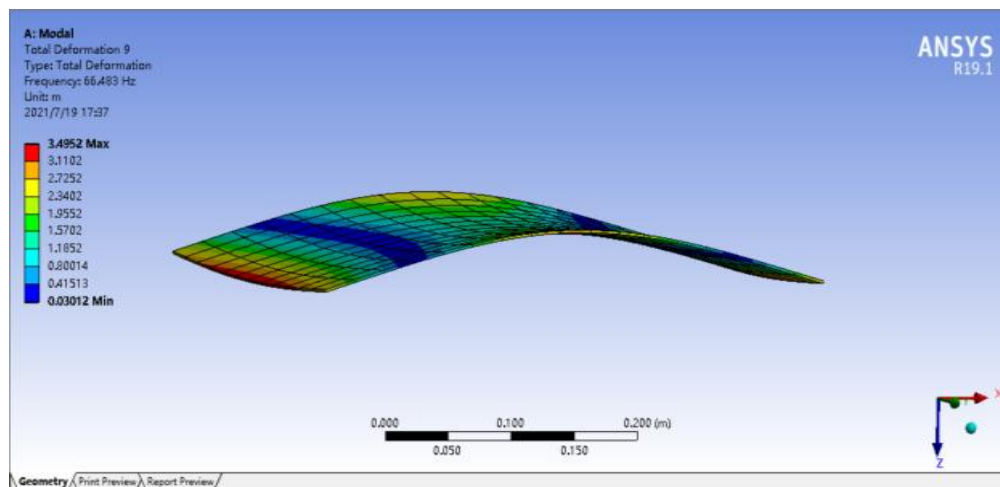


Fig. 10. The first-order bending mode of the computational modal analysis

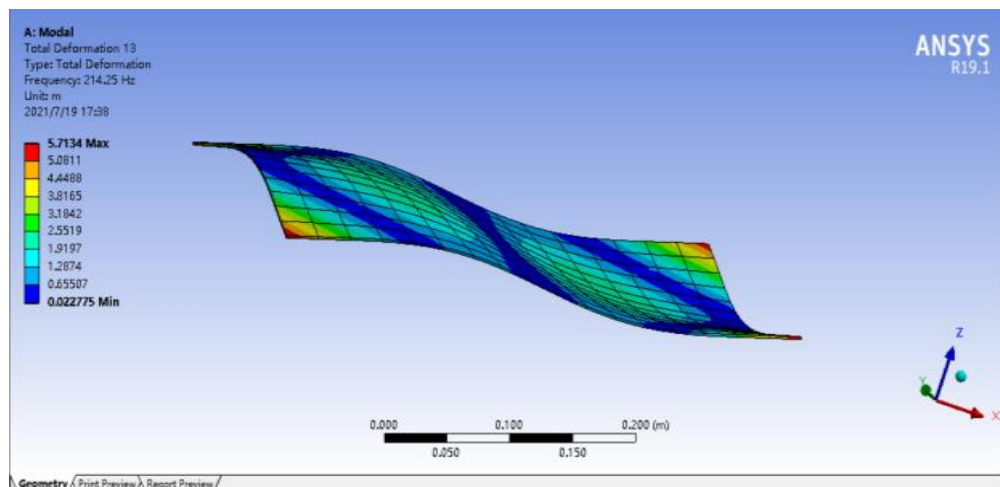


Fig. 11. The second-order bending mode of the computational modal analysis

**Table 4.** Computational Modal Natural Frequency

Content	First-order Torsional Mode	First-order Bending Mode
Mode frequency (Hz)	56.19	66.48

**Test Modal and Computational Modal Results and Analysis**

It is known from the calculation results that the test model was consistent with the computational model, and the corresponding mode frequency values are shown in Table 5.

**Table 5.** Computational Model and Test Model Results

Parameter	Test Model	Computational Model	Error Frequency
Length (mm)	430.5	430.5	-
Width (mm)	301.5	301.5	-
Thick (mm)	1.88	1.88	-
Density (kg/m <sup>3</sup> )	1604.4	1604.4	-
First-order torsional mode (Hz)	53.94	56.19	4%
First-order bending mode (Hz)	66.24	66.48	0%
The elastic modulus (E) (MPa)	64731	65207	0.7%
The shear modulus (G) (MPa)	22160	24044	7.8%
Poisson's ratio ( $\mu$ )	0.46	0.46	-

As the natural frequency is the primary parameter of vibration characteristics and the natural characteristics of the object, it is only related to the stiffness, mass, and overall dimensions of the object. Therefore, the natural frequency can be used as the strength index of the object under certain conditions for the detection of mechanical properties (Guo 2013).

According to Table 5, the error frequencies of the E and G values calculated by the computational modal and test modal were 0.7% and 7.8%, respectively, which indicated that the two analysis results had a high degree of coupling (Wang and Ghanem 2021; Zhou *et al.* 2021).

**CONCLUSIONS**

1. Based on the test modal analysis and computational modal analysis of carbon fiber boards, the elastic modulus, shear modulus, and Poisson's ratio of a carbon fiber board can be obtained accurately and reliably.
2. The computational modal analysis was consistent with the first-order bending and first-order torsional of the test modal analysis; the frequency error was within 5%, and the error rates of E and G were 0.7% and 7.8%, respectively. This showed that the coupling degree of the two analysis results were high, and the correctness and reliability of the mode shape and elastic constant of carbon fiber boards can be well realized.
3. At the same time, the research was conducive to strengthening the use of finite element computational modal analysis and better promote its application in the nondestructive

testing and quality grading of carbon fiber boards, which has engineering application value for improving the optimal design of the dynamic characteristics of carbon fiber board and promoting the faster development of the wood structure market. At the same time, it provides convenience for the dynamic testing and research of the elastic constants of wood-carbon fiber composites in the future.

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